

**Flight Design general aviation GmbH**

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**SB-ASTM-CTSW-16**

**SB-ASTM-CTLS-17**

**SB-CTLS-ELA-001**

**SB-ASTM-MC\_-04**

**SB-LTUL-CTSW-15**

**SB-LTUL-CTLS-15**

**Revision 00**

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# Service Bulletin

## Repair procedure to additional wires protection

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**SB-CTLS-ELA-001**

**SB-ASTM-MC\_-04**

**SB-LTUL-CTSW-15**

**SB-LTUL-CTLS-15**

## Repeating Symbols:

Please pay attention to the following symbols throughout this document emphasizing particular information.

- ▲ **Warning:** Identifies an instruction, which if not followed may cause serious injury or even death.
- **Caution:** Denotes an instruction which if not followed, may severely damage the aircraft or could lead to suspension of warranty.
- **Note:** Information useful for better handling.

## 1 Planning Information

### 1.1 Affected Aircraft

Type: CT and MC  
 Model: CTSW, CTLS, CTLS-ELA, CTSW-LSA, CTLS-LSA, MC  
 Serial Number: CTSW:  
 E-18-10-55.  
 CTLS:  
 F-18-10-52, F-18-10-54, F-18-11-03, F-18-11-04, F-19-03-01,  
 F-19-04-51, F-19-04-55, F-19-05-01, F-19-05-02, F-19-05-05,  
 F-19-05-06, F-19-08-52, F-19-11-01, F-19-11-02, F-19-11-04.  
 CTLS-ELA:  
 F-18-08-51, F-18-08-52, F-18-11-05, F-19-08-51, F-18-09-01.

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CTSW-LSA:

D-19-05-03.

CTLS-LSA:

F-18-07-02, F-18-07-03, F-18-03-01, F-18-03-02, F-18-09-02,  
F-18-09-05, F-18-10-50.

MC:

A-19-01-06.

Applicable Countries: Not limited

**1.2 Concurrent Documents**

- none -

**1.3 Reason**

An electrical short circuit occurred due to damage of the wire insulation that lead to contact of the wire to the steel part connected to electrical ground.

**1.4 Subject**

Repair procedure to additional wires protection for aircraft owners.

**1.5 Compliance**

This Service Bulletin requires immediate action.

Compliance must be shown within the next 10 hours.

**▲Warning:** Non-compliance with these instructions could result in further damages, personal injuries or death.

**1.6 Approval**

The technical content of this document is approved under the authority of DOA ref. EASA.21J.416

**1.7 Type of Maintenance**

Line

**1.8 Personnel Qualifications**

Aircraft owner, case dependent supported by an aircraft inspector as per national regulations.

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**1.9 Release to Service**

Conduct of this SB must be inspected by an aircraft inspector according to the national applicable regulations for the country of registry of the aircraft.

**1.10 Weight and Balance**

Not applicable

**1.11 References**

Not applicable

**1.12 Superseded Documents**

- none -

**1.13 Contact Details**

For further information on conduct of this SB, or to report any Safety of Flight or Service Difficulty issues contact your Distributor responsible for your country. Your Distributor can be located via the Flight Design general aviation GmbH website: [www.flightdesign.com](http://www.flightdesign.com) under "Dealer Location".

In cases where the local distributor is not known or available contact Flight Design general aviation GmbH directly: [customer.care@flightdesign.com](mailto:customer.care@flightdesign.com)

Specific contact in USA:

Flight Design USA

P.O. Box 325 South Woodstock, CT 06267

Tel: 860 963 7272 / Fax: 860 963 7152

Web: [www.flightdesignUSA.com](http://www.flightdesignUSA.com)

E-Mail: [airworthiness@flightdesignUSA.com](mailto:airworthiness@flightdesignUSA.com)

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**1.14 Disclaimer**

This Service Bulletin has been generated with utmost care. Nevertheless errors and misunderstandings can never be fully excluded. In case of any doubts the applicant of this Service Bulletin is requested to contact Flight Design immediately to clarify the issue.

**2 Resources**

-none-

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-none-

**2.2 Parts**

-none-

**2.3 Materials**

Additional wires protection set.

**2.4 Tools**

Allen key 3 mm

Philips screwdriver

**2.5 Special Tools**

-none-

**2.6 Manpower**

The described task can be performed within approximately 1 hours (working time).

**2.7 Cost**

Not applicable

**3 Instructions****3.1 General**

Inspect wires carefully for an indication of wear on the insulation before you wrap the wires with the tape.

This chapter provides extended explanations for the sequential repair steps. See in the Table.

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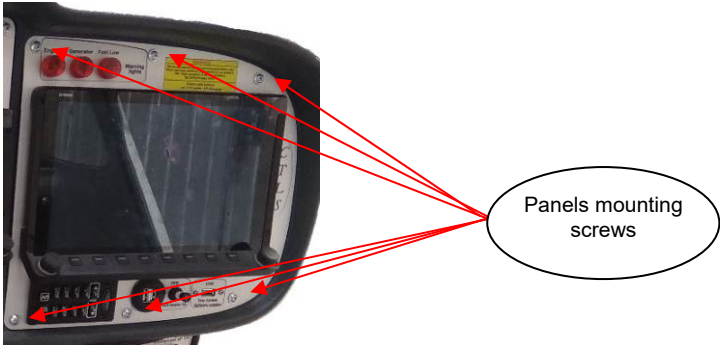
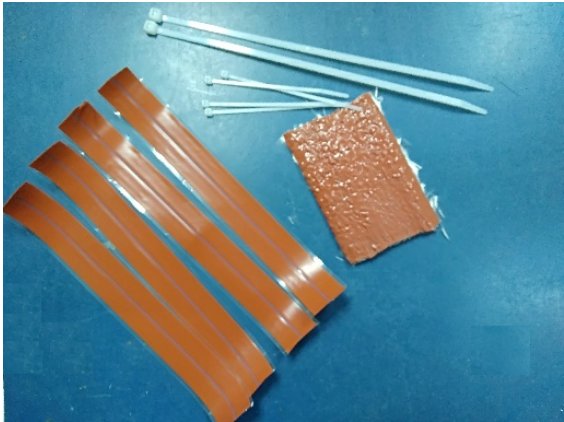

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Step	Action	Reference
1	<p>Remove Panel Right section. Unscrew panel mounting screws.</p> 	
2	<p>Open additional wires protection set.</p> 	
3	<p>Pull out the wires as much as possible.</p> 	

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4

Remove the protective film from pyrosil tape. Wind the pyrosil tape around the wires\*.

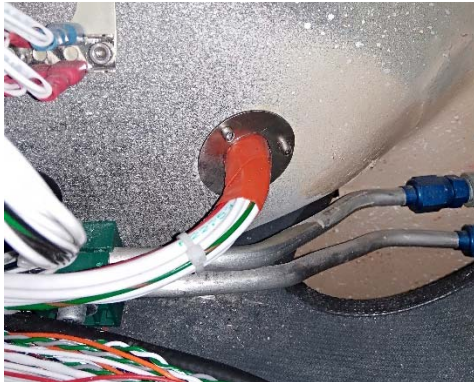


\*Installation: Stretch tape during installation using at least a 50% overlap. For best results, use on clean, dry surfaces free from dirt and oil.

The protective layer must be at least 5 layers.

5

Return the wires to their original position.



6

Remove upper cowling.



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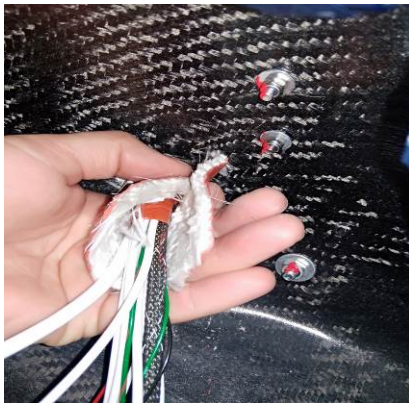
**SB-LTUL-CTLS-14**

**Revision 00**

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Wrap fire protection hose around flange and wires. Tighten TY-wraps around flange and wires.



7

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8	<p>Remove the protective film from pyrosil tape. Wind the pyrosil tape around the wires*.</p>  <p>*Installation: Stretch tape during installation using at least a 50% overlap. For best results, use on clean, dry surfaces free from dirt and oil.</p>	
9	Install Panel Right section and upper cowling.	

### 3.2 Documentation

Conduct of this SB must be logged in the aircraft log book with date and signature of the responsible Person conducting the SB. National regulations have to be considered.

## 4 Appendix

### 4.1 Changes to Previous Revision

Original Issue – no changes

### 4.2 Feedback Template

For further information or of Flight Service Difficulty issues FLIGHT DESIGN general aviation GmbH directly: [airworthiness@flightdesign.com](mailto:airworthiness@flightdesign.com) or [info@flightdesign.com](mailto:info@flightdesign.com). Flight Design will assist solve any problem with your aircraft.



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This form has to be used to coordinate with Flight Design the required spare parts and to communicate the completion of the SB listed in the header of this page.

Submission can be done by mail, Fax or as scanned copy by e-mail to Flight Design GmbH (see header of this page) or to your national Flight Design Distributor who will forward the information for you.

### Information to obtain spare parts

Aircraft Type: CT  
 Model: CT\_\_  
 Serial Number: \_\_\_\_\_  
 Aircraft Owner, Name and Address: \_\_\_\_\_  
 \_\_\_\_\_  
 \_\_\_\_\_

Check question 1 Yes  No

Check Question 2 Yes  No

### Confirmation of Completion

Conduct of this SB has been done at the following workshop:

\_\_\_\_\_  
 Responsible Mechanic:  
 \_\_\_\_\_

Date: \_\_\_\_\_ Location: \_\_\_\_\_ Signature: \_\_\_\_\_

**Warning:** As long as the form has not been submitted to Flight Design, this SB is considered not complied-with for the respective aircraft.